

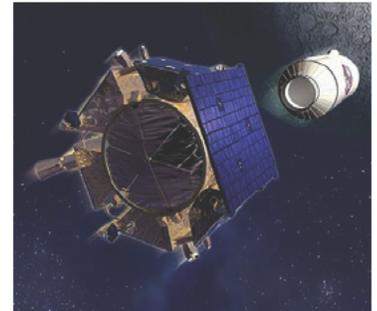
COMET

COMMERCIAL ESPA TUG



The COMET is a propulsive tug for secondary payload deployment based on the Moog EELV Secondary Payload Adapter (ESPA) ring. The COMET is designed to launch on EELV-class launch vehicles (Atlas V, Delta IV, Falcon 9) while supporting up to 1500 kg of secondary payloads with a variety of standard separation systems. The COMET can be used to disperse small satellite constellations, act as a Hosted

Payload platform, or deliver a single spacecraft to its ideal orbit. The COMET has its own avionics, power, propulsion, and communications systems that are configurable for short durations up through multi-year missions in a wide range of orbits.



COMET

The COMET enables small satellites to launch on rideshare missions and achieve their ideal orbit with minimal affect on the primary spacecraft.

POTENTIAL LAUNCH CONFIGURATIONS

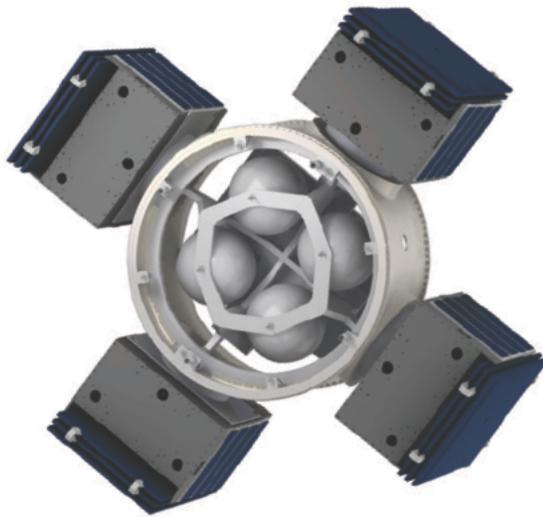
- Below a primary payload
- As part of a multi-manifest stack with other adapters and spacecraft
- As an upper stage or tug for a primary spacecraft or stack of ESPA rings

MISSION APPLICATIONS

- Secondary Payload Tug
- Hosted Payload Platform
- Technology Demo Platform
- Comm Relay for Small Sat applications
- Science Mission Platform

VERTICAL INTEGRATION

The COMET draws from Moog's extensive experience in structures, spacecraft avionics, in-space propulsion,



SPECIFICATIONS

Parameter	Performance
Total Impulse	349 kN-sec (78,500 lbf – sec)
Thrust (Nominal)	4 x 22 N (4x5 lbf)
Propellant Type	Hydrazine (4:1 Blowdown)
C&DH + EPS	Integrated Avionics Unit (IAU)
Attitude Control	3-Axis Stabilized Control < ±1° Knowledge < ±0.1°
Communications	User Configurable Baseline X-band up to 100 Mbps
Orbital Lifetime	Up to 3 years in LEO
Dry Mass (w/out Payloads)	348 kg (767 lbm)
Wet Mass (w/out Payloads)	501 kg (1105 lbm)
Payload Mass	Up to 1500 kg (3307 lbm) depending on port size and quality
Nominal ESPA Ring Height	≥32" (42" ESPA Grande option)
Payload Interface	12xØ11.7", 6xØ15", 4xØ24", 5xØ24", Customizable Four Point Mount & Others Available
Example Mission Scenario	Total Payload Mass: 5 x 300 kg – 1500 kg Total Delta-V: 180 m/s CONOPS 1.) COMET drop-off at 450 km SSO → Release 300 kg payload 2.) COMET maneuvers to 550 km SSO → Release 300 kg payload 3.) COMET maneuvers to 650 km SSO → Releases three (3) payloads 4.) COMET de-orbit burn to 250 km x 650 km



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